

Scientific challenges for the development of **hybrid modeling / optimization / control** strategies of complex fluid systems

Chair PROVE (Green Aeronautical Propulsion)

Denis Sipp, Angelo Iollo

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Co-chairs: D. Sipp & A. Iollo

- **Collaboration between ONERA and Région Nouvelle Aquitaine (academic & industry)**
- **Subject: Innovative methods to design low-emission propulsion (hybrid modeling, optimization and control strategies)**
- **Funding for PhD (6+), Post-docs (2+) and Interns (10+)**
- **Partners:**
 - **Research centers: ONERA, CEA CESTA**
 - **Academic: Bordeaux (MEMPHIS, CARDAMOM), Poitiers (PPRIME), Bidart (ESTIA)**
 - **Industry: Safran HE, Safran T, Ingeliance, AKIRA**

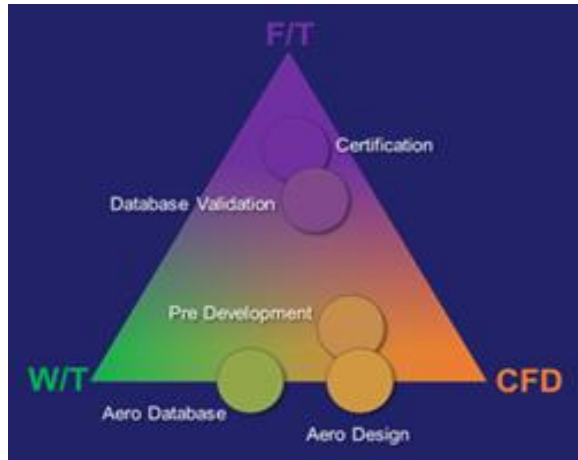
Outline

- 1) **Context and objectives of the PROVE chair**
- 2) **Scientific challenges**
- 3) **Partnership**

The digital transformation of the aerospace industry

The case of aerodynamics

Today



Objective



« Mix of data » in various phases of aircraft development. Left: today, Right: tomorrow

Emerging opportunities for predictive CFD for off-design commercial airplane flight characteristics, J. Slotnick/Boeing, G. Heller/Airbus, 54th A3F, 25-27 mars 2019

Digitalisation of aeronautical industry

Setting up digital twins

- **Numerical models** accompanying the development of an aircraft from the pre-project phase to operation and withdrawal
- **Multi-fidelity** and **evolutionary** models
 - Quick "low-fidelity" models at the pre-project stage
 - Accurate "high-fidelity" models for the later phases
- **Hybrid models** built by combining
 - Physical models (conservation principles)
 - Data from various sources (engine test benches, numerical simulations, flight tests, etc.) and characterized by different uncertainties
- AI techniques to manage the **mass of data**
- **Modeling, Optimization and Control** techniques (objective=low emissions) under constraints (high level of flight safety)

Outline

- 1) Context and objectives of the PROVE chair
- 2) Scientific challenges
- 3) Partnership

Scientific Challenges

- Challenge 1: **Input to Output** ROM with large-dimensional input space
- Challenge 2: **Input to State** ROM with few CFD evaluations
- Challenge 3: **Data-based Turbulence Modeling** for accurate and fast CFD computations
- Challenge 4: **State reconstruction** with sparse measurements
- Challenge 5: **Dynamic ROM** with finite-amplitude perturbations

Challenge 1: Input to Output ROM with large-dimensional input space

- **Objective:** considering the optimisation problem, $\min_p J(p)$, adapt response surface-based optimization tools to take into account the gradient information $\nabla_p J$, routinely provided by new CFD codes.

- **State of the art:**

- Surrogate models (Kriging)
- Clustering (Gaussian mixtures)
- Design of experiment (DoE), adaptive enrichment (EM)
- Reduction of the dimension of the input space (Active sub-spaces U_1 , P. Constantine):

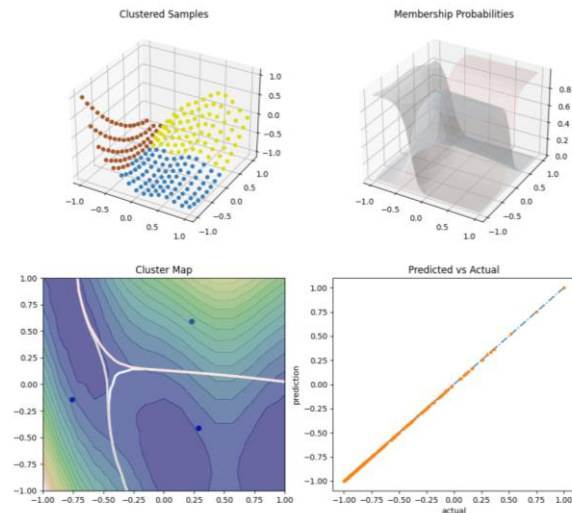
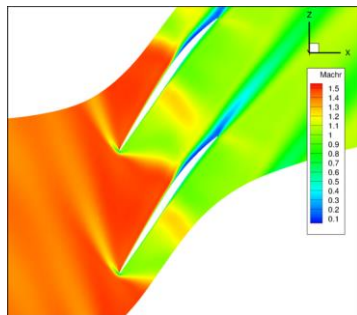
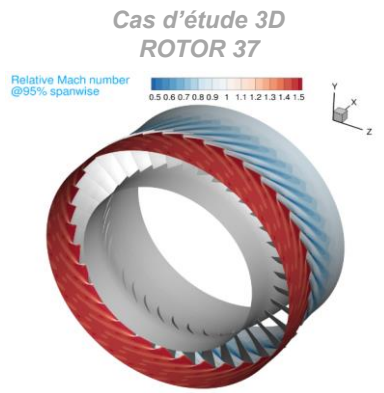
$$(\nabla_{p_1} J \quad \nabla_{p_2} J \quad \cdots) = (U_1 \quad U_2) \begin{pmatrix} \Sigma_1 & 0 \\ 0 & \approx 0 \end{pmatrix} (V_1 \quad V_2)^* \approx U_1 \Sigma_1 V_1^*$$

$$J(p) = J(U_1[U_1^* p] + U_2[U_2^* p]) \approx J(U_1[U_1^* p])$$

- **People involved:** M. Chapron (PhD student), Ch. Blondeau (DAAA/MSAE), I. Salah El Din (DAAA/H2T), M. Bergmann (MEMPHIS)
- **Related project :** UE NEXTAIR
- **References:**
 - Constantine, P. G., Dow, E., & Wang, Q. (2014). Active subspace methods in theory and practice: applications to kriging surfaces. *SIAM Journal on Scientific Computing*, 36(4), A1500-A1524.
 - Bettebghor, D., Bartoli, N., Grihon, S., Morlier, J., & Samuelides, M. (2011). Surrogate modeling approximation using a mixture of experts based on EM joint estimation. *Structural and multidisciplinary optimization*, 43(2), 243-259.
 - Bouhlel, M. A., Hwang, J. T., Bartoli, N., Lafage, R., Morlier, J., & Martins, J. R. (2019). A Python surrogate modeling framework with derivatives. *Advances in Engineering Software*, 135, 102662.

Challenge 1: Input to Output ROM with large-dimensional input space

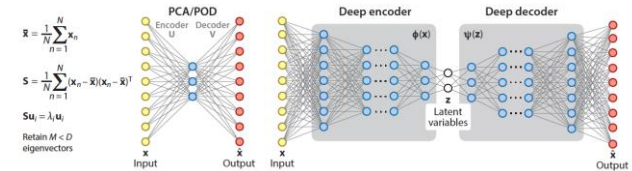
- **Scientific challenge:** efficient exploration and compact representation of the surrogate $J(p)$ when p is high dimensional (in an industrial process one can typically have 100 to 10000 parameters!) in the case where there are several local minima (multi-modality)
- **Approach considered:** Combination of the "Active Subspace" technique due to the availability of $J(p)$ and $\nabla_p J$ in new CFD solvers + clustering (Gaussian mixture, EM)



- **Expected results:**
 - Methodology for surrogate models and robust optimization in the case of a very high dimensional input space (≈ 10000)
 - Implementation of the clustered Active Subspaces approach in SMT

Challenge 2: Input to State ROM with few CFD evaluations

- **Industrial need:** in the design phase, need to optimize a very large number of parameters (the shape of a blade) with objectives on the state of the system (suppression of a vortex for example)
- **Objective:** From realizations of the state as a function of parameters, identify a **subspace** and build a **reduced-order model** representing the input - state - output relationships.
- **State of the art:**
 - Identification of a reduced dimensional subspace in the output space:
 - POD
 - Auto-Encoder
 - Reduced-Order model:
 - Physics-based : Galerkin projection, DEIM, hyper-reduction, ...
 - Data-based : Clustering, GP, ML, SINDY, ...

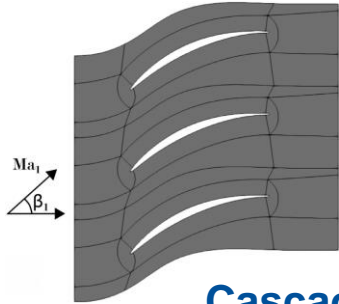


- **People involved:** J. Labatut (thésard), J.-B. Chapelier (DAAA/NFLU), A. Remigi (ST), T. Taddei (IMB/MEMPHIS)
- **Related projects:** UE ARIA (A. Iollo)
- **References :7**

- **Iollo, A., & Taddei, T. (2022).** Mapping of coherent structures in parameterized flows by learning optimal transportation with Gaussian models. *Journal of Computational Physics*, 471, 111671.
- **Chapelier, J. B., De La Llave Plata, M., Renac, F., & Lamballais, E. (2014).** Evaluation of a high-order discontinuous Galerkin method for the DNS of turbulent flows. *Computers & Fluids*, 95, 210-226.
- **Sipp, D., de Pando, M. F., & Schmid, P. J. (2020).** Nonlinear model reduction: a comparison between POD-Galerkin and POD-DEIM methods. *Computers & Fluids*, 208, 104628.
- **Cordesse, P., Remigi, A., Duret, B., Murrone, A., Ménard, T., Demoulin, F. X., & Massot, M. (2020).** Validation strategy of reduced-order two-fluid flow models based on a hierarchy of direct numerical simulations. *Flow, Turbulence and Combustion*, 105(4), 1381-1411.

Challenge 2: Input to State ROM with few CFD evaluations

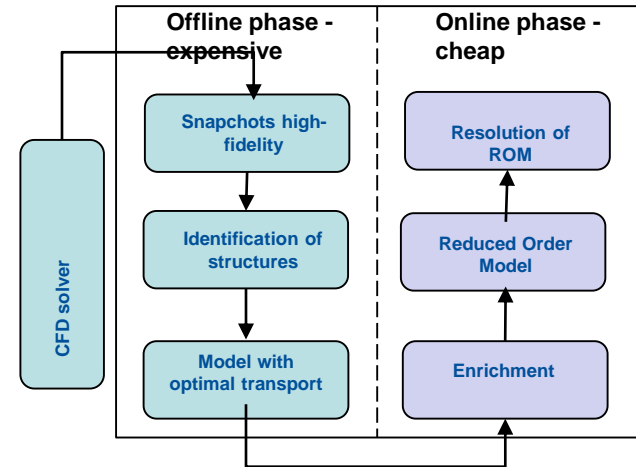
- **Scientific challenge** : compression of a compact structure (vortex) that moves (depending on the parameters), taking into account the boundary conditions during the optimal transport phase, feature selection, enrichment of the basis, ...
- **Approach considered**: non-linear interpolation (registration-based)
- **Test case**: reduction of the DG Aghora code on the following data set



Cascade V103

Flow conditions for various compressor cascade scenarios.

Scenario	S_1	S_2	S_3	S_4
β_1	36.99°	39.97°	44.09°	49.2°
Ma_1	0.654	0.674	0.666	0.65
Re_1	302K	302K	298K	289K
Tu (%)	2.9	3.4	3.4	3.5

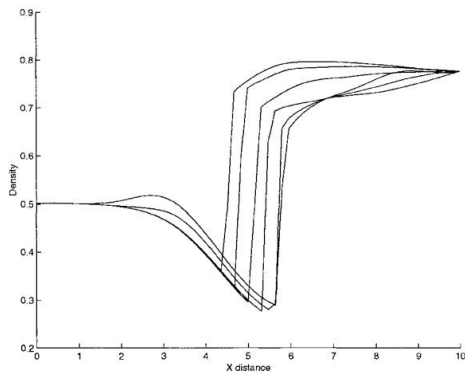
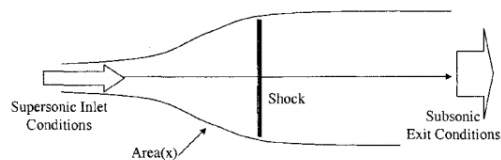


Expected results:

- Methodology for setting up a parameterized ROM
- Model for a rotor-stator channel, multi-channel coupling for the representation of an entire stage, then multi-stage coupling for the representation of a compressor (Post-Doc)
- Module for model reduction in a python library

Challenge 2: Input to State ROM with few CFD evaluations

Parametrized nozzle flow: $p_{out} \in [0,1]$

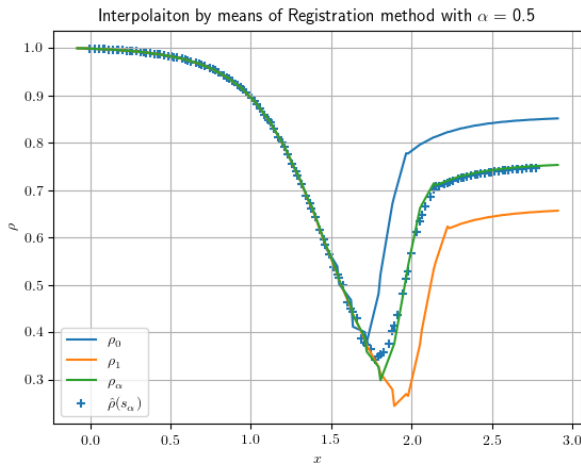


CFD solver
 $\hat{\rho}(p_{out}, x)$

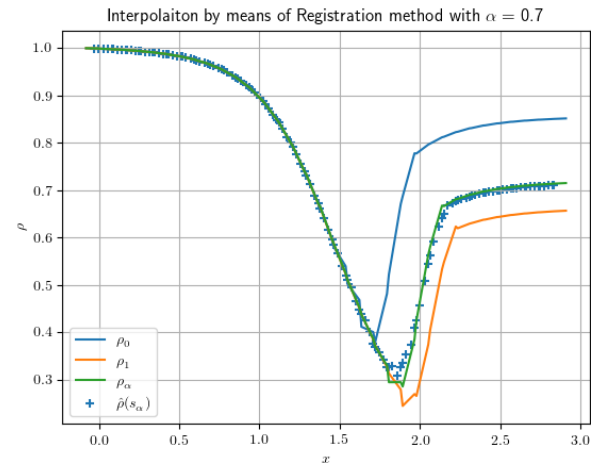
Two solutions for $\rho_0(p_0 = 0.85, x)$ and $\rho_1(p_1 = 0.65, x)$.

Approximation: $\hat{\rho}(s_\alpha, x)$ which gives a prediction of the solution at low cost.

$$\text{Relative error: } \varepsilon_\alpha = \frac{\|\hat{\rho}_{s_\alpha} - \rho_{obj}\|_2^2}{\|\rho_{obj}\|_2^2}$$



$\varepsilon_\alpha = 0.013$
 $p_{out} = 0.75$



$\varepsilon_\alpha = 0.017$
 $p_{out} = 0.71$

Challenge 3: Data-based Turbulence Modeling for accurate and fast CFD computations

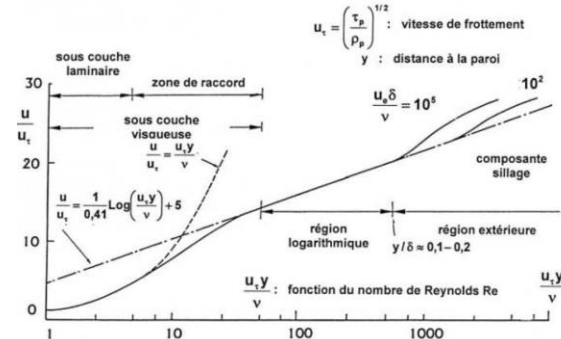
- **Objective:** improve fidelity and reduce the cost of RANS aerodynamic simulations used in design offices

- **State of the art:**

- Improvement of turbulence closure models
- Improvement of wall laws
- Empirical wall laws : Musker, Afzal ...

- **Model for internal region:**

$$\frac{\bar{u}(y)}{u_\tau} = f\left(\frac{\rho_w u_\tau}{\mu_w} y, \frac{\mu_w}{\rho_w^2 u_\tau^3} \partial_x \bar{p}\right), u_\tau = \sqrt{\frac{\mu_w \partial_y \bar{u}|_{y=0}}{\rho_w}}$$



- **People involved:** M. Romanelli (PhD) S. Beneddine (ONERA/MASH), I. Mary (ONERA/DEFI), H. Beaugendre (INRIA/CARDAMOM), M. Bergmann (INRIA/MEMPHIS)

- **Related projects:** ANR NEWMAC (with SAFRAN TECH and DLR)

- **References:**

- **Beneddine, S.** (2022). Nonlinear input feature reduction for data-based physical modeling. arXiv preprint arXiv:2206.07400.
- Constant, B., Péron, S., **Beaugendre, H.**, & Benoit, C. (2021). An improved immersed boundary method for turbulent flow simulations on Cartesian grids. *Journal of Computational Physics*, 435, 110240.
- Péron, S., Benoit, C., Renaud, T., & **Mary, I.** (2021). An immersed boundary method on Cartesian adaptive grids for the simulation of compressible flows around arbitrary geometries. *Engineering with Computers*, 37(3), 2419-2437.
- Volpiani, P. S., Meyer, M., Franceschini, L., Dandois, J., Renac, F., Martin, E., ... & **Sipp, D.** (2021). Machine learning-augmented turbulence modeling for RANS simulations of massively separated flows. *Physical Review Fluids*, 6(6), 064607.

Challenge 3: Data-based Turbulence Modeling for accurate and fast CFD computations

● Approach followed:

- Steady RANS resolved up to the wall=reference
- Identification of inputs by physical approach and by data-based approach (mutual information, determination of the most influential parameters)

$$\frac{\bar{u}(y)}{u_\tau} = f\left(\frac{\rho_w u_\tau}{\mu_w} y, \frac{\mu_w}{\rho_w^2 u_\tau^3} \partial_x \bar{p}, \text{Compressibility, Geometry, Non-locality, ...}\right)$$

$$\bar{v}(y) = ???$$

$$\bar{\rho}(y) = ???$$

$$\bar{v}_t(y) = ???$$

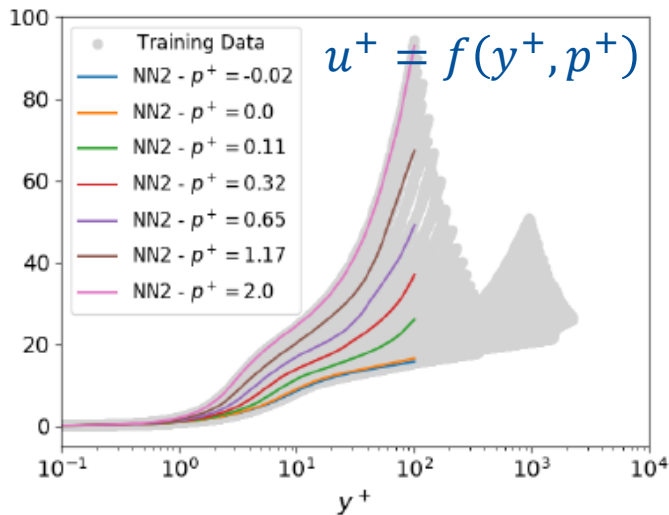
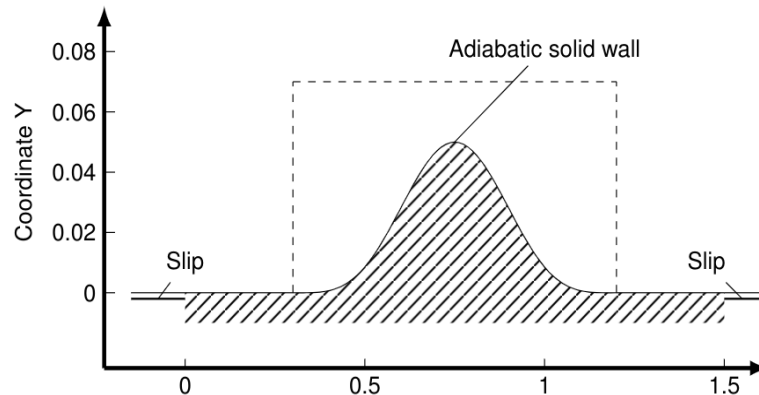
2. GAN (image based approach) ?
 3. Separated flows, LES data or experimental data-base
- Unsteady LES resolved up to the wall=reference
 - Modeling of fluctuations

● Expected results

- More accurate and more robust laws of the wall for RANS and LES simulations
- Data-based methodology to determine best input / output features (Mutual information)

Challenge 3: Data-based Turbulence Modeling for accurate and fast CFD computations

- High-fidelity database: 2D RANS
- Bump flows: $Re \in [10^6, 10^7]$, $h/L \in [0,05; 0,08]$
- Moderate pressure gradients without separation



RANS interface at $y^+ \approx 10$					
	$h = 0$	$h = 0.05$	$h = 0.06$	$h = 0.07$	$h = 0.08$
$Re = 1 \cdot 10^7$			0.72%		0.88%
$Re = 8 \cdot 10^6$			0.75%		
$Re = 6 \cdot 10^6$		0.77%	0.8%	0.85%	
$Re = 3 \cdot 10^6$	0.61%	0.84%	0.93%	0.92%	
$Re = 1 \cdot 10^6$		1.26%	1.35%	1.54%	

RANS interface at $y^+ \approx 30$					
	$h = 0$	$h = 0.05$	$h = 0.06$	$h = 0.07$	$h = 0.08$
$Re = 1 \cdot 10^7$			1.68%		2.09%
$Re = 8 \cdot 10^6$			1.92%		
$Re = 6 \cdot 10^6$		1.94%	2.27%	2.53%	
$Re = 3 \cdot 10^6$	1.21%	2.94%	3.41%	3.76%	3.98%
$Re = 1 \cdot 10^6$		5.54%	6.28%	6.84%	

Learning

Challenge 4: State reconstruction with sparse measurements

- **Industrial need:** reconstruction of the flow in a confined system with moving parts, in which measurements are difficult and scarce
- **Objective:** to be able, from sparse measurements, to reconstruct the large-scale state of a system
- **State of the art:**

$$\min_{\mathbf{w}} \|\mathbf{y}(\mathbf{w}) - \bar{\mathbf{y}}\|_{M^{-1}}^2 + \|\mathbf{w} - \bar{\mathbf{w}}\|_{F^{-1}}^2$$

- Ensemblistic methods (EnKF): non-intrusive, very expensive, covariances OK, generation of ensemble members?
- Variational methods: intrusive, expensive, covariances KO
- Nudging : almost non-intrusive, not expensive, covariances KO
- **Persons involved :** R. Villiers (thésard), V. Mons (DAAA/MAPE), E. Lamballais (PPRIME/Poitiers), M. Meldi (LMFL/Lille)
- **References :**
 - **Mons, V.**, Chassaing, J. C., Gomez, T., & Sagaut, P. (2016). Reconstruction of unsteady viscous flows using data assimilation schemes. *Journal of Computational Physics*, 316, 255-280.
 - **Meldi, M.**, & Poux, A. (2017). A reduced order model based on Kalman filtering for sequential data assimilation of turbulent flows. *Journal of Computational Physics*, 347, 207-234.
 - Dairay, T., **Lamballais, E.**, Laizet, S., & Vassilicos, J. C. (2017). Numerical dissipation vs. subgrid-scale modelling for large eddy simulation. *Journal of Computational Physics*, 337, 252-274.
 - Franceschini, L., **Sipp, D.**, & Marquet, O. (2020). Mean-flow data assimilation based on minimal correction of turbulence models: Application to turbulent high Reynolds number backward-facing step. *Physical Review Fluids*, 5(9), 094603.

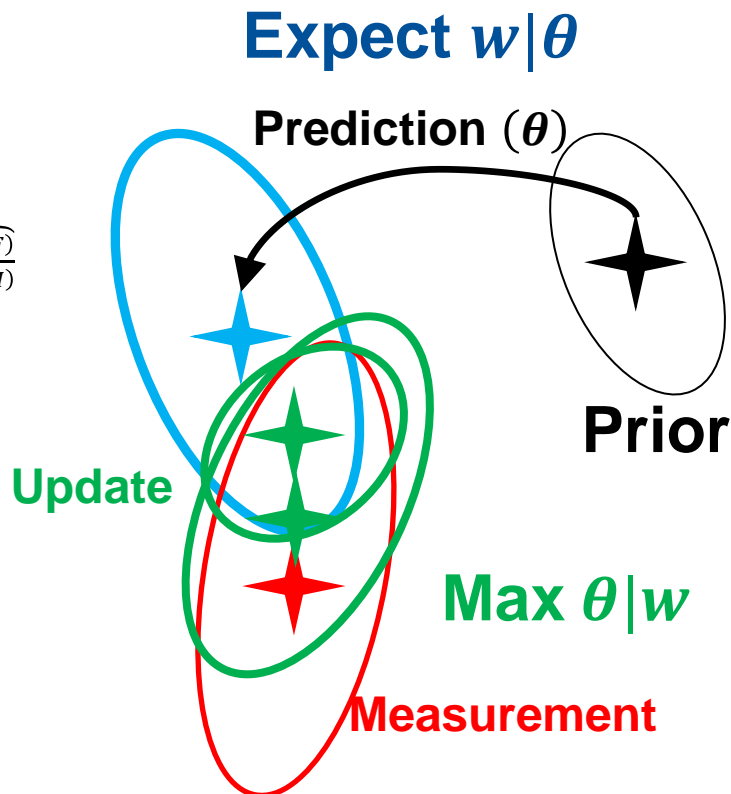
Challenge 4: State reconstruction with sparse measurements

- **Scientific challenge:** best combine statistical information from sparse observations with an imperfect predictive model to be improved
- **Approach considered:** Expectation-Maximization (EM) algorithm provides the best estimate of the state w (ensemble-based method), parameters θ and state-covariance F (machine learning)

Maximisation of probability likelihood: $p(w, \theta, F | y, M) = p(y | w, M) p(w | \theta, F) \frac{p(w, F)}{p(y, M)}$

$$p(y | w, M) = \exp\left(\frac{-\frac{1}{2}\|y - y(w)\|_M^2}{(2\pi)^{N_y | M|}}\right) \quad p(w | \theta, F) = \exp\left(\frac{-\frac{1}{2}\|w - w(\theta)\|_F^2}{\sqrt{(2\pi)^{N_w | F|}}}\right)$$

- **Reference:** Bocquet, M., Brajard, J., Carrassi, A., & Bertino, L. (2020). Bayesian inference of chaotic dynamics by merging data assimilation, machine learning and expectation-maximization. *Foundations of Data Science*, 2(1), 55.
- **Test-cases:** unsteady turbulent flows (turbulent boundary layer, cylinder at $Re=12000$), then ROTOR 37
- **Expected results:**
 - Methodology for estimating (w, θ, F)
 - Improved URANS model and LES subgrid model correction with explicit filtering



Challenge 4: State reconstruction with sparse measurements

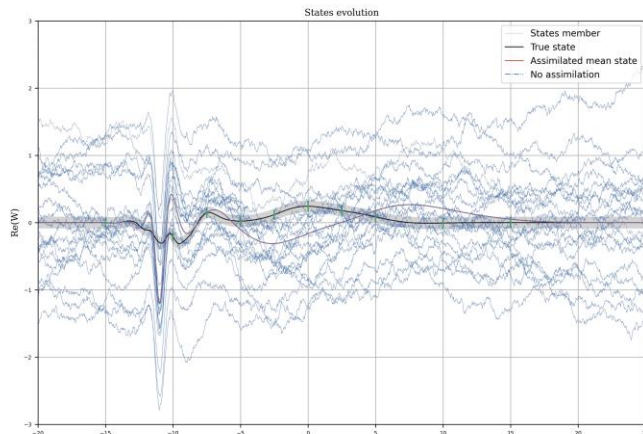
1D model $w(x, t)$ (which mimicks NS) :

$$\partial_t w + U \partial_x w + w|w|^2 = \left(\mu_0 - c_u^2 + \frac{\mu_2 x^2}{2} \right) w + \nu \partial_{xx} w + \eta(x, t)$$

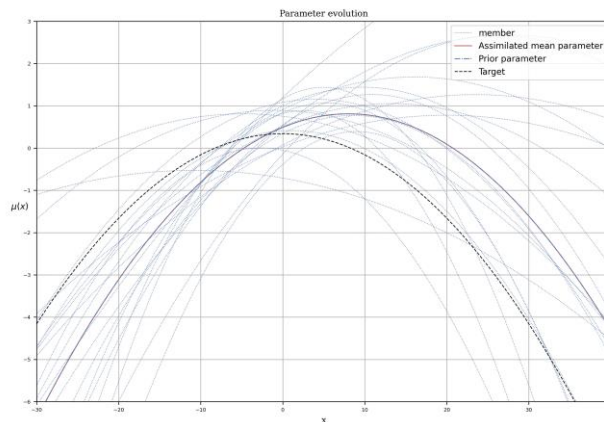
Imperfect model : $\partial_t w + U \partial_x w = \mu_\theta(x) w + \nu \partial_{xx} w$

Measures : $w(x_i, t)$

Assimilation technique : Sequential Kalman



State reconstruction $w(x, t)$



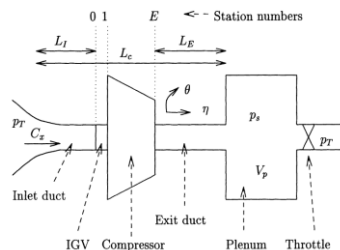
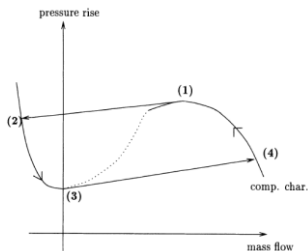
Parameter reconstruction $\mu_\theta(x)$

Challenge 5: Dynamic ROM with finite-amplitude perturbations

- **Industrial need:** in a compressor, monitoring and control of the pumping phenomenon (strongly non-linear)
- **Objective** From sparse measurements, build a predictive model of the state of a system (with strong nonlinearities) and control it
- **State of the art:**
 - Reduced-order models:
 - Model-based linearized approaches: BPOD, résolvant
 - Data-based linearized approaches: Identification, Loewner (Ch. Poussot-Vassal), ...
 - Hybrid nonlinear approaches: POD+Projection and variants (L. Cordier), coupling of physical models and data (APHYNITY=>N. Thome, P. Gallinari, CD-ROM=>A. Bucci, L. Mathelin, M. Schoenauer)
 - Data-based nonlinear approaches : Koopman, DMD, Reservoir-computing (L. Magri)
 - Linear control : synthèse robuste Hinf (P. Apkarian), MPC, ...
 - Joint learning of model & control with data: RL (Mathelin, Cordier, Beneddine, thèse R. Paris)
 - Alternative: iterative procedure with robust linear controllers based on linear models (Leclercq et al, 2019)
- **People involved:** PhD student (?), C. Leclercq (DAAA/MASH), L. Cordier (PPRIME), L. Magri (Imperial College), J. Marty (DAAA/H2T)
- **Related projects:** PhD of N. Lepage with N. Thome & I. Mortazavi by combining a physical model and data
- **References:**
 - Leclercq, C., Demourant, F., Poussot-Vassal, C., & Sipp, D. (2019). Linear iterative method for closed-loop control of quasiperiodic flows. *Journal of Fluid Mechanics*, 868, 26-65.
 - Bucci, M. A., Semeraro, O., Allauzen, A., Wisniewski, G., Cordier, L., & Mathelin, L. (2019). Control of chaotic systems by deep reinforcement learning. *Proceedings of the Royal Society A*, 475(2231), 20190351.
 - Menier, E., Bucci, M. A., Yagoubi, M., Mathelin, L., & Schoenauer, M. (2022). CD-ROM: complementary deep-reduced order model. *arXiv preprint arXiv:2202.10746*.
 - Yin, Y., Le Guen, V., Dona, J., de Bézenac, E., Ayed, I., Thome, N., & Gallinari, P. (2021). Augmenting physical models with deep networks for complex dynamics forecasting. *Journal of Statistical Mechanics: Theory and Experiment*, 2021(12), 124012.

Challenge 5: Dynamic ROM with finite-amplitude perturbations

- **Scientific challenge:** control of bifurcations (nonlinearity)
- **Approach considered:**
 - ✓ Representation of the surge phenomenon in the framework of dynamic systems



$$\dot{\Psi} = \frac{W/H}{4B^2} \left(\frac{\Phi}{W} - \frac{1}{W} \Phi_T(\Psi) \right) \frac{H}{l_c} \quad (1.16)$$

$$\dot{\Phi} = \frac{H}{l_c} \left(-\frac{\Psi - \psi_{c0}}{H} - \frac{1}{2} \left(\frac{\Phi}{W} - 1 \right)^3 + 1 + \frac{3}{2} \left(\frac{\Phi}{W} - 1 \right) \left(1 - \frac{J}{2} \right) \right)$$

$$\dot{J} = J \left(1 - \left(\frac{\Phi}{W} - 1 \right)^2 - \frac{J}{4} \right) \varrho,$$

Plenum pressure coeff.

Annulus average mass flow coeff.

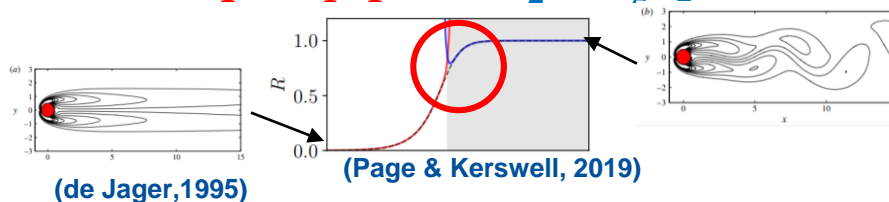
Rotating stall amplitude squared

Moore & Greitzer (1986)

- ✓ Linearized model obtained by DMDc (fully data-driven) to replace the solver (partially data and physics based)
- ✓ Learning a nonlinearity by machine learning to augment a linear DMDc model (Khodkar et al. 2021)

$$\dot{x}_1 \approx A_1 x_1$$

$$\dot{x}_2 \approx A_2 x_2$$



(de Jager, 1995)

(Page & Kerswell, 2019)

$$\dot{x} = Ax + f(x)$$

- ✓ Predictive control, adaptive control

Cross-over problem between two DMD representations

Challenge 5: **Dynamic ROM** with finite-amplitude perturbations

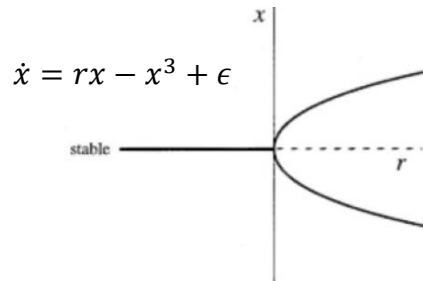
- **Expected results:**

- Representation of the surge phenomenon as a dynamic system
- Data-based control methodology to detect and suppress surge
- Module in a software library for the identification and control of a bifurcation

Challenge 5: Dynamic ROM with finite-amplitude perturbations

Normal form of a supercritical fork bifurcation

5 ARX composite models capable of tracking ~30% of branch changes



$$\dot{x} = rx - x^3 + \epsilon$$

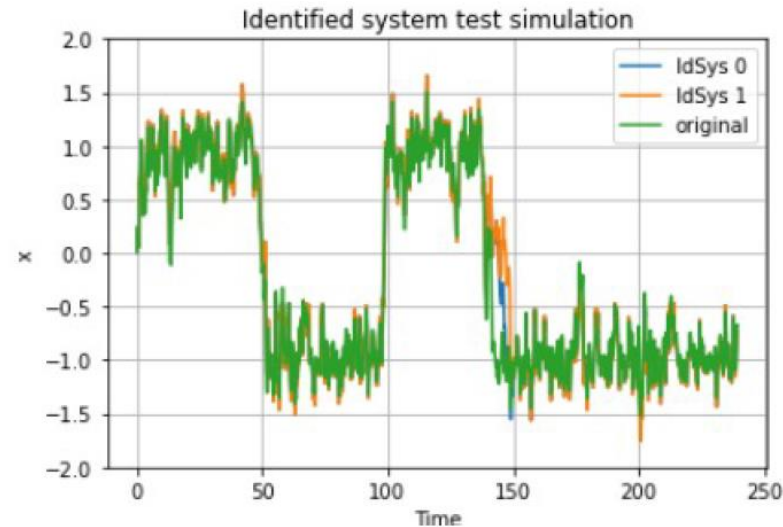
$$x_{k+1} = \left[\sum_{i=0}^{n_a} a_i \sqrt{\mu} x_{k-i} + \sum_{i=0}^{n_b} b_i \sqrt{\mu} u_{k-i} + \left(\sum_{i=0}^{n_a} a_i \sqrt{\mu} - 1 \right) \sqrt{\mu} \right] \mathbb{1}_{[0.73 < x_k]} +$$

$$\left[\sum_{i=0}^{n_a} a_i \sqrt{\mu/3} x_{k-i} + \sum_{i=0}^{n_b} b_i \sqrt{\mu/3} u_{k-i} + \left(\sum_{i=0}^{n_a} a_i \sqrt{\mu/3} - 1 \right) \sqrt{\mu/3} + \frac{2}{3} \sqrt{\frac{\mu}{3}} \Delta t \right] \mathbb{1}_{[0.6 < x_k \leq 0.73]} +$$

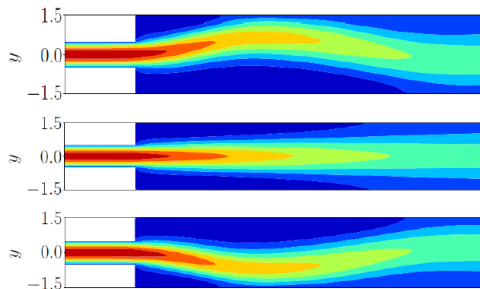
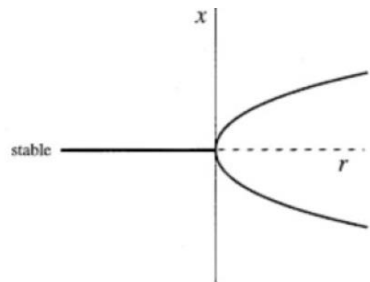
$$\left[\sum_{i=0}^{n_a} a_i^0 x_{k-i} + \sum_{i=0}^{n_b} b_i^0 u_{k-i} \right] \mathbb{1}_{[-0.6 \leq x_k \leq 0.6]} +$$

$$\left[\sum_{i=0}^{n_a} a_i^{-\sqrt{\mu/3}} x_{k-i} + \sum_{i=0}^{n_b} b_i^{-\sqrt{\mu/3}} u_{k-i} - \left(\sum_{i=0}^{n_a} a_i^{-\sqrt{\mu/3}} - 1 \right) \sqrt{\mu/3} - \frac{2}{3} \sqrt{\frac{\mu}{3}} \Delta t \right] \mathbb{1}_{[-0.73 \leq x_k < -0.6]} +$$

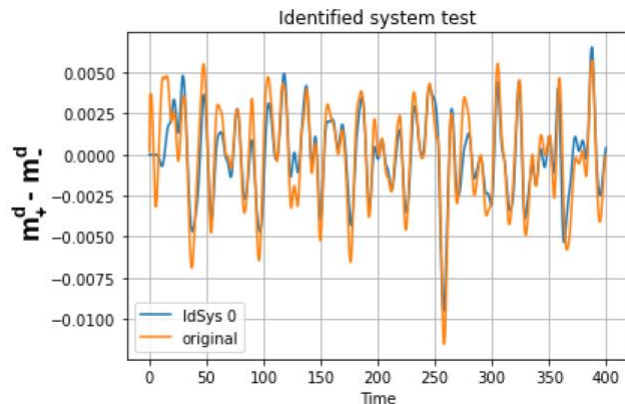
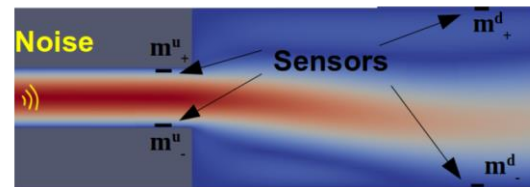
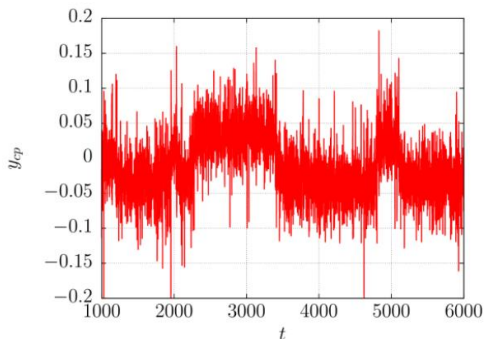
$$\left[\sum_{i=0}^{n_a} a_i^{-\sqrt{\mu}} x_{k-i} + \sum_{i=0}^{n_b} b_i^{-\sqrt{\mu}} u_{k-i} - \left(\sum_{i=0}^{n_a} a_i^{-\sqrt{\mu}} - 1 \right) \sqrt{\mu} \right] \mathbb{1}_{[x_k < -0.73]}$$



Challenge 5: Dynamic ROM with finite-amplitude perturbations



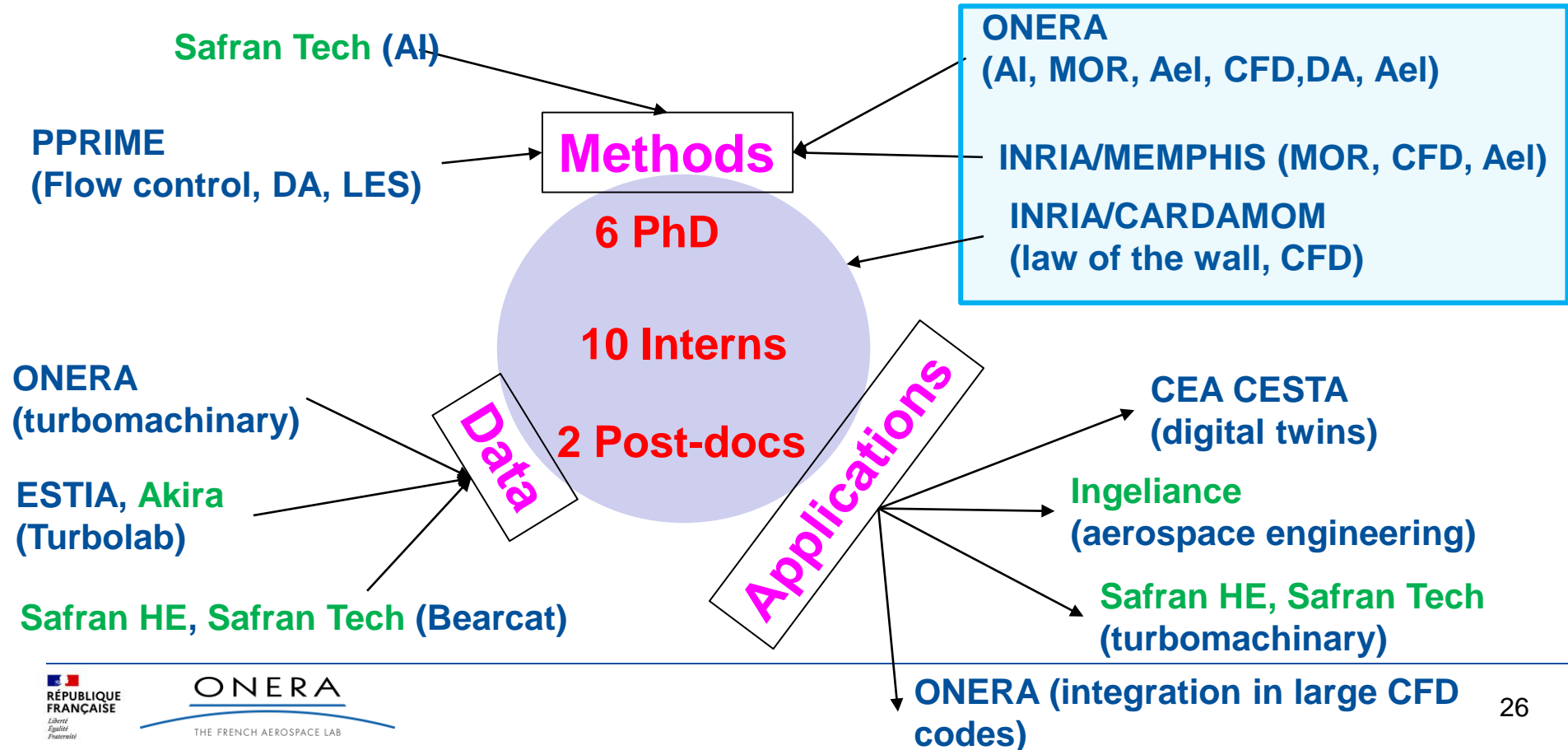
+ finite amplitude noise



Outline

- 1) Context and objectives of the PROVE chair
- 2) Scientific challenges
- 3) Partnership

PROVE Chair (co-funded by Région Nouvelle-Aquitaine)



Data from representative experimental test-benches

TURBOLAB (ESTIA, AKIRA)

TURBOLAB une plateforme technologique pour la propulsion innovante

Des moyens d'essais complets ...

... et des véhicules d'essais



Emulateur Batterie
250kW / 5-1000V DC
Alimentation bidirectionnelle
Réinjection réseau



Groupe d'eau glacée
AERMEC ANL 202
50kW



**Servitude Carburant /
Hydrogène**
Pompes de régulation
H2 stockage 200 bars
Système complet



Des locaux adaptables
Ateliers, salle de réunion,
salles de formation



Salle de contrôle
Pilote, Surveillance,
Caméras



Acquisition
Instrumentation spécifique
Centre d'acquisition
équipier et polyvalente



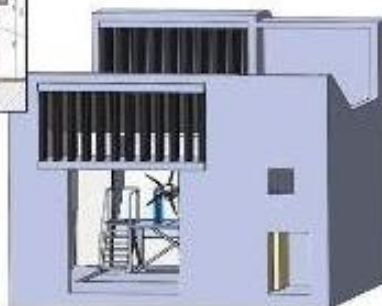
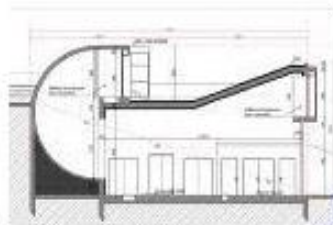
Turbofan DGEN380
250 daN de poussée



e-TurboProp OPUS380-H
230kW de puissance sur
arbre, thermique et
électrique



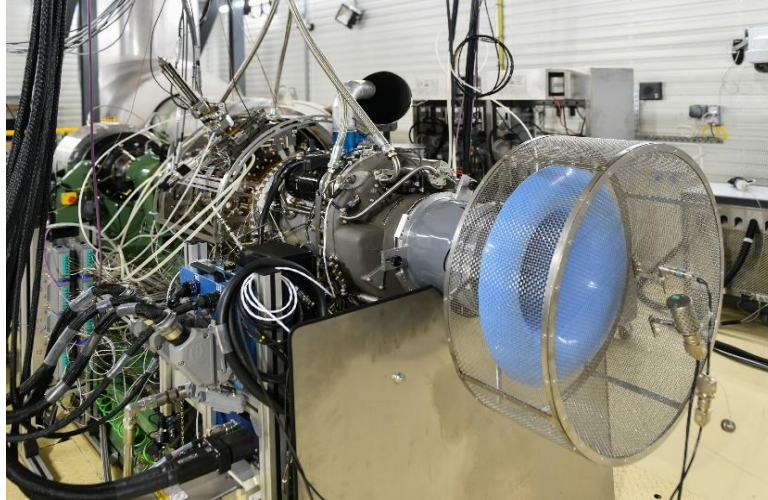
Banc hélice
Moteur 250kW, mesure
des efforts/couples



Data from representative experimental test-benches

BEARCAT (Safran HE & Safran Tech)

Le banc « BEARCAT » (Banc d'Essai Avancé pour la Recherche en Combustion et Aérothermique des Turbomachines) est un turbomoteur basé sur le moteur MAKILA. Il est dédié à la caractérisation fine des phénomènes aérothermiques se produisant dans la chambre de combustion et la turbine haute pression ainsi qu'à leurs interactions.



BEARCAT

People involved

DAAA:

- MAAA : Denis Sipp
- MAPE : Vincent Mons, Olivier Marquet
- MASH : Samir Beneddine, Colin Leclercq
- H2T : Itham Salah el Din, Julien Marty, Raphaël Barrier
- DEFI : Ivan Mary
- MSAE : Christophe Blondeau
- NFLU : Jean-Baptiste Chapelier, Florent Renac
- ACI : Pedro Stefanin Volpiani

DTIS:

- IVA : F. Champagnat

MEMPHIS :

- Angelo Iollo
- Michel Bergmann
- Tommaso Taddei

CARDAMOM :

- Heloise Beaugendre

PPRIME :

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- Laurent Cordier
- Lionel Agostini
- Guillaume Lehnasch
- Nassim Razaaly
- Vincent Jaunet

LMFL :

- Marcello Meldi

CEA CESTA :

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- Sébastien Esquieu

SafranT :

- Alberto Remigi
- Grégory Dergham
- Alessandro Bucci

SafranHE :

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- Franck Matrippolito

Ingéliance :

- Matthieu Puyo
- Guillaume Ruiz
- Mathias Truel

PhD students:

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- Raphael Villiers
- Maxime Chapron
- Jon Labatut
- Bartolomeo Fanizza
- Thomas Philipert

Thank you for your attention